



بِسْمِ اللَّهِ الرَّحْمَنِ الرَّحِيمِ

**MINISTRY OF TRANSPORT AND COMMUNICATIONS
CIVIL AVIATION DEPARTMENT**

Male'
Republic of Maldives

AIRWORTHINESS DIRECTIVE

No. CAD/DHC-6/4
21 September 1995

APPLICABILITY

De Havilland DHC-6 "Twin Otter" aircraft
Serial No. All

SUBJECT

Inspection for cracks in the fuselage frames

REFERENCE

Airworthiness Directive CF-95-12 (Transport Canada); dated 11 July 1995

REASONS

There has been reported instances of cracking on the skin flange of the machined fuselage frame at fuselage section FS 239.

ACTION AND COMPLIANCE

Action to be accomplished as follows:

Inspect the fuselage frame for cracks by visual method per Accomplishment Instructions described in the De Havilland Service Bulletin (SB) 6/521, dated 05 July 1995, or later revision.

- 1.0 If crack(s) are found which are within the tolerances specified in paragraph 5, Accomplishment Instructions of SB 6/521, perform repetitive inspections in accordance with the De Havilland Inspection requirement Manual, PSM 1-6-7 Temporary Revision 77, or as detailed in PSM 1-6-7 normal revision No.8, when issued.
- 2.0 If crack(s) are found which exceed the tolerances specified in paragraph 5, Accomplish Instructions of SB 6/521, repair or replace the subject part area before further flight.

EFFECTIVITY

This Directive becomes effective on 21 September 1995

Abdul Razzak Idris
DIRECTOR OF CIVIL AVIATION